



Goal

Our team is attempting to build a remote controlled aircraft powered by solar cells. We hope to use solar power to produce run times significantly better than those of traditional, battery-powered aircraft.

Background

The purpose of this project is to design an aircraft that integrates solar cells into design for sustained flight. Transportation accounted for more than one third of CO₂ emissions, and fossil fuel consumption continues to increase globally so alternatives fuels would have a tremendous impact on the environment.

Many engineers hope to create an airplane that is powered with solar powered, but solar cells were too heavy and inefficient to make this goal plausible. With new developments in the semiconductor industry, new solar cells offer promise for applications that were previously not possible such as the NASA Helios, an aircraft meant to stay aloft indefinitely under purely solar power.

Applications for solar powered airplanes range from long term aerial surveillance to planning for construction and monitoring crops. In more recent years, advanced solar cells have become more accessible to the common consumer. With this project, we hope to use newly available technologies to create a smaller scale solar powered aircraft.

Team Members		
Member / Major	Subteam Responsibilities	
Alvin Cao / ME & MSE	 Solar cell and airfoils research, poster 	
 Joseph Cuevas / ME & MSE 	 Solar cell research, fuselage design 	
Christopher Gay / ME & MSE	 Team leader, components and overall design research 	
 Kenny Miranda / ME & MSE 	 Materials and dihedrals research 	
 Jonathan Wong / ME & MSE 	 Materials research, poster 	

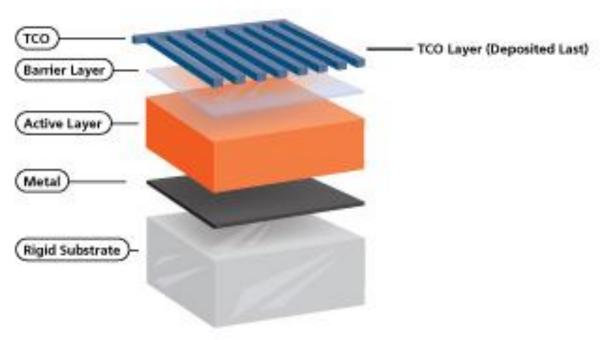
Solar Powered Airplane

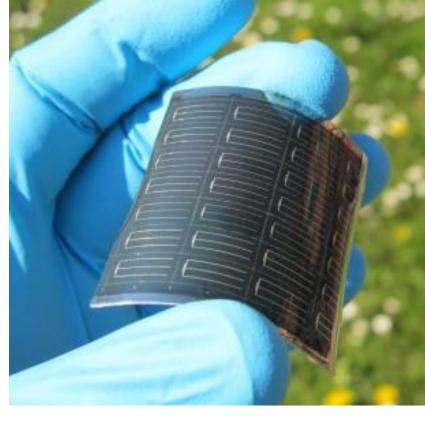
Alvin Cao, Joseph Cuevas, Christopher Gay, Kenneth Miranda, Jonathan Wong **Department of Materials Science** Professor Albert Yee

The Henry Samueli School of Engineering

CIGs Solar Cells

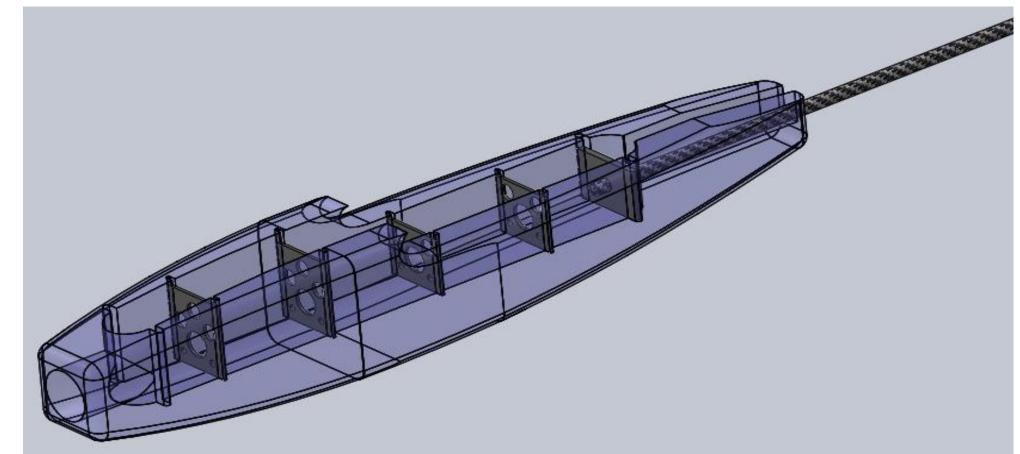
Voltage	• 18.5V	Pros:	
• Current	• 243mA (0.234A)	Completely flexible	ne
• Voltage (oc)	• 25.5V	Easy to connect	
• Current (sc)	• 295mA (0295A)	Inexpensive	Hi
Power Output	 4.5 [W] → .00769 [W/cm²] 	Direct exposure to sunlight; don't need to rely on	(2 Hi
• Total Size	• (170 x 344mm)	transparent film	Ca
Weight	 (105 g) → .180 [g/cm²] 	Cons:	Th Th
• Price	 \$40.00 ea → 0.0684 [\$/cm²] 	Moderately heavy Does not come in smaller	gr Th
• Efficiency	• ~20%	sizes	fo
CIGS		51205	(2





Fuselage

Our preliminary fuselage design incorporates a streamlined teardrop shape in an attempt to minimize drag. The carbon fiber boom used to support the tail wings will be a low profile, low weight, high strength alternative to having a foam fuselage that extends to the end.



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Project definition, planning, budget Fall 2012: Initial research for design and materials selection Submit project to UROP, Fall Design Review





Materials and Estimated Cost

The following table sums up the most important materials we will need for the project and their costs (including tax and shipping):

ltem	Cost (USD)
Optic 6 Radio Transmitter and Receiver System	\$240
ec HS-85MG+ Servos	\$75
HG2025 Inrunner Brushless Motor	\$75
Creations Phoenix 25 Electronic Speed Controller	\$90
er Power TP13003SPP45 Battery	\$45
ler Power TP610C Lithium Battery Charger (for testing and d use)	\$140
ler Power TP425C light Lithium Charger (to be dismantled board battery charging circuit)	\$70
/aveSol Mobile (EIPV Module) – WSME-0045 Solar Cells	\$880
r 9 Heat Gun (for shrinking films onto the plane)	\$30
r 80W Soldering Iron (for connecting solar cells and g in field repairs)	\$50
ute Epoxy (for securing major joints and making in field s)	\$50
Cure Cyanoacrylate Glue (fast and minor repairs)	\$20
film rolls	\$50
yrene blocks	\$50
n fiber rods	\$70
reinforced Nylon sheets	\$400

Progress and Future Work

- <u>Winter 2013</u>: Detailed design & Implementation plan, revised budget Initial prototype, initial test, update design approach E-Week BBQ & Senior Display
- Spring 2013: Design Revision, Final Prototype & Demonstration Final Design Review & Symposium

Contact Information

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